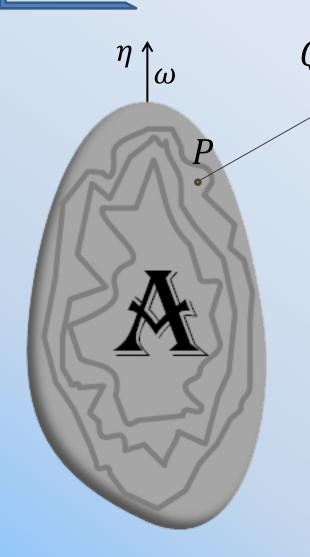
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DYNAMICS OF SPACE ELEVATOR ON ASTEROID

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ASTEROID WITH ATTACHED PENDULUM

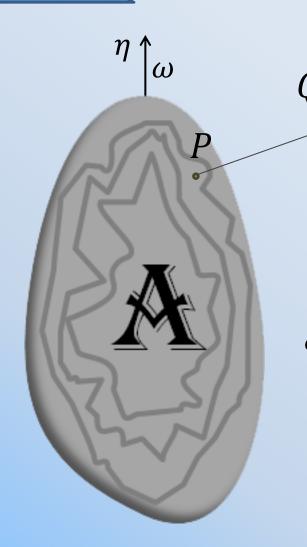


✓ Suppose PQ = l

- ✓ Introduce a frame Oζξη (RF) uniformly rotating with the body about the axis Oη.
- ✓ This rotations is assumed to be stable.
- ✓ Suppose RF is composed by the asteroid's central principal axes of inertia.
- ✓ Positions of points P and Q with respect to RF are given by they radius-vectors:

$$\mathbf{OP} = (p_z, p_x, p_y)$$
 and $\mathbf{OQ} = (z, x, y)$.

AUGMENTED POTENTIAL AND ROUTH FUNCTION



The potential of centrifugal force is

$$U_C = -\frac{1}{2}m\omega^2(z^2 + x^2)$$

The gravitational potential depends on mass distribution of asteroid A. Denote it as

$$U_N = U_N(z, x, y).$$

Constraint:

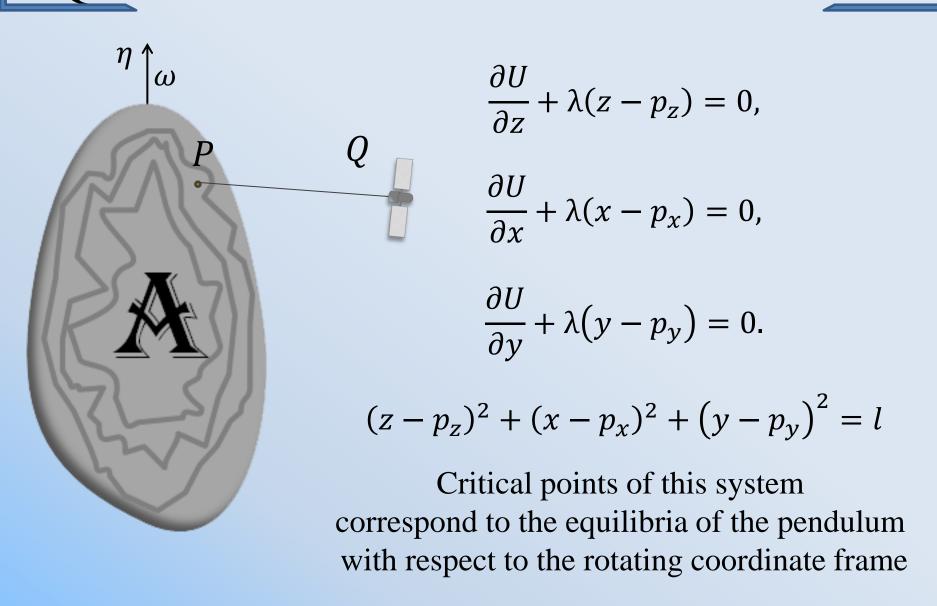
$$\varphi = (z - p_z)^2 + (x - p_x)^2 + (y - p_y)^2 - l = 0$$

Write down the Routh function:

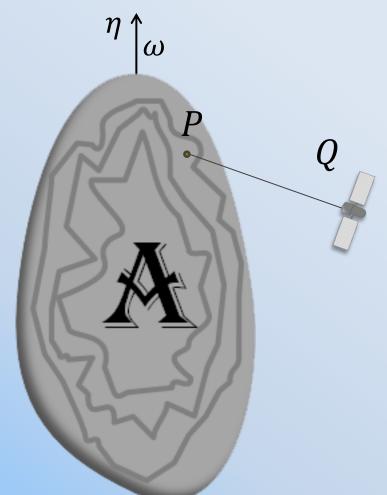
$$W(x, y, z, \lambda, \omega, l) = U + \frac{\lambda}{2}\varphi$$

where
$$U = U_C + U_N$$

EQUILIBRIA POINTS OF THE PENDULUM



EXCLUDING LAGRANGIAN MULTIPLIER



$$\frac{\partial U}{\partial x}(y-p_y) - \frac{\partial U}{\partial y}(x-p_x) = 0,$$

$$\frac{\partial U}{\partial y}(z-p_z) - \frac{\partial U}{\partial z}(y-p_y) = 0,$$

$$\frac{\partial U}{\partial z}(x - p_x) - \frac{\partial U}{\partial x}(z - p_z) = 0,$$

It expresses that the moment of force relative to the point *P* is equal to zero.

POSSIBLE EQUILIBRIA OF THE SPACECRAFT

$$\frac{\partial U}{\partial x}(y - p_y) - \frac{\partial U}{\partial y}(x - p_x) = 0, \quad \checkmark \text{ Suppose anchor } P \text{ is fixed.}$$

$$\checkmark \Gamma \text{ corresponds to possible}$$

$$\frac{\partial U}{\partial y}(z - p_z) - \frac{\partial U}{\partial z}(y - p_y) = 0,$$

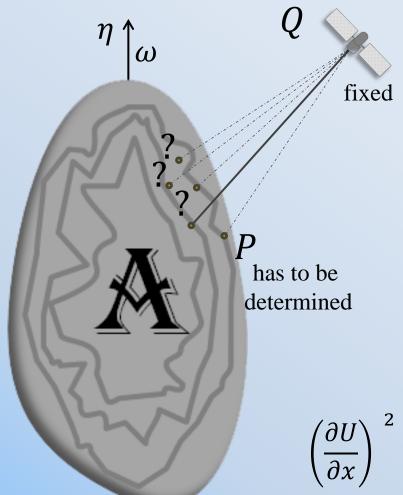
$$\frac{\partial U}{\partial z}(x - p_x) - \frac{\partial U}{\partial x}(z - p_z) = 0,$$

- equilibrium of point Q
- ✓ Curve Γ passes through the anchor Pand points where

$$\frac{\partial U}{\partial x} = \frac{\partial U}{\partial y} = \frac{\partial U}{\partial z} = 0$$

 \checkmark When spacecraft Q is located in the libration point, then the tether is slack: $\lambda = 0$.

SOLUTION OF THE INVERSE PROBLEM



 \checkmark Suppose l is pre-defined

The problem is to determine the anchor point for a given length of the tether if spacecraft *Q* is in equilibrium in a given point of the rotating reference frame.

From system of equations for equilibria of the pendulum one can obtain:

$$\left(\frac{\partial U}{\partial x}\right)^{2} + \left(\frac{\partial U}{\partial y}\right)^{2} + \left(\frac{\partial U}{\partial z}\right)^{2} =$$

$$= \lambda^{2} \left((z - p_{z})^{2} + (x - p_{x})^{2} + (y - p_{y})^{2} \right)$$

Taking into account constraint $\varphi = 0$, one arrives at $\lambda = \pm l^{-1} |\operatorname{grad} U|$

SOLUTION OF THE INVERSE PROBLEM

✓ Suppose
$$\lambda \neq 0$$

From system of equations for equilibria of the pendulum one can obtain:

$$\begin{cases} \frac{\partial U}{\partial z} + \lambda(z - p_z) = 0 \\ \frac{\partial U}{\partial z} + \lambda(z - p_z) = 0 \\ \frac{\partial U}{\partial z} + \lambda(z - p_z) = 0 \end{cases} \Rightarrow p_x = x + \frac{l}{|\text{grad}U|} \frac{\partial U}{\partial x}$$
$$p_y = y + \frac{l}{|\text{grad}U|} \frac{\partial U}{\partial y}$$

Thus the location of the anchor is found

POSITION OF ANCHOR ON THE ASTEROID SURFACE

- ✓ Take into account that the anchor should be located on the asteroid surface.
- \checkmark Let this surface be described in the $O\zeta\xi\eta$ as $f(p_z, p_x, p_y) = 0$
- \checkmark Suppose Q is fixed (for a while)

$$p_z = z + \frac{l}{|\text{grad}U|} \frac{\partial U}{\partial z}; \ p_x = x + \frac{l}{|\text{grad}U|} \frac{\partial U}{\partial x}; \ p_y = y + \frac{l}{|\text{grad}U|} \frac{\partial U}{\partial y}$$

The system is not self-contained:

3 equations for determination 4 unknowns: p_z , p_x , p_y and l

The equation of the surface $f(p_z, p_x, p_y) = 0$ closes this system.

SPECIAL CASE OF ASTEROID SURFACE

Suppose that asteroid surface is a sphere of radius *R* centered in *O*:

$$p_z^2 + p_x^2 + p_y^2 - R^2 = 0$$

$$p_z = z + \frac{l}{|\operatorname{grad} U|} \frac{\partial U}{\partial z}; \ p_x = x + \frac{l}{|\operatorname{grad} U|} \frac{\partial U}{\partial x}; \ p_y = y + \frac{l}{|\operatorname{grad} U|} \frac{\partial U}{\partial y}$$

$$\lambda = l^{-1}|\operatorname{grad} U|$$

$$\Rightarrow p_z^2 + p_x^2 + p_y^2 - R^2 = 0$$

$$(z + ln_z)^2 + (x + ln_x)^2 + (y + ln_y)^2 = R^2$$

where
$$(n_z, n_x, n_y) = \frac{\text{grad}u}{|\text{grad}u|}$$

SPECIAL CASE OF ASTEROID SURFACE

Suppose that the pre-defined equilibrium does not coincide with a libration point: $|\text{grad}U| \neq 0$.

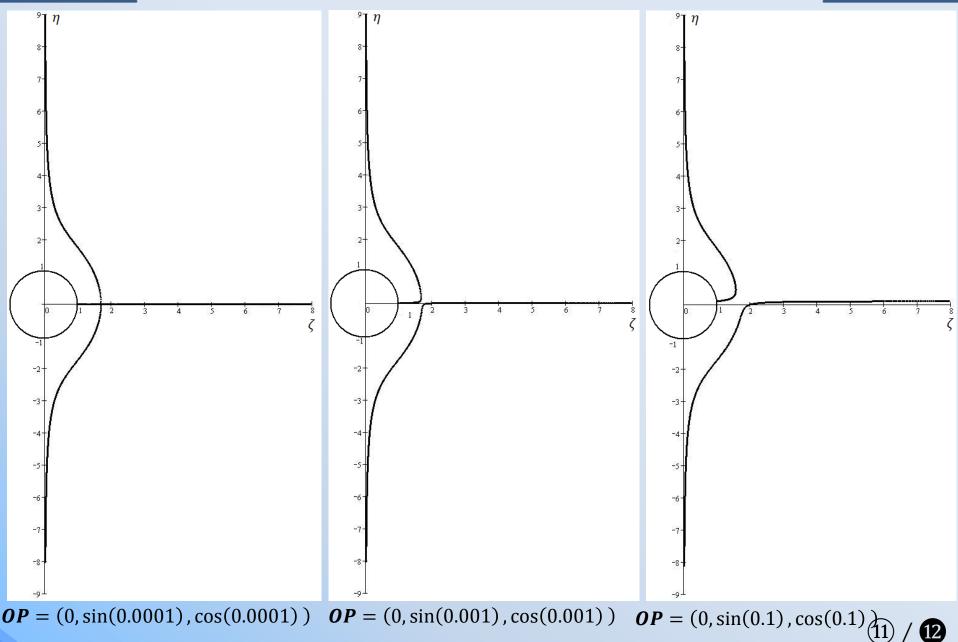
For its discriminant *D* the following inequality holds true:

$$\frac{D}{4} = p_1^2 - p_0 = (zn_z + xn_x + yn_y)^2 - (z^2 + x^2 + y^2 - R^2) =$$

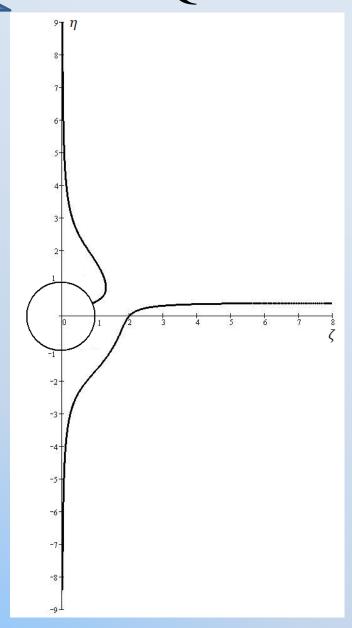
$$= (n_x y - n_y x)^2 + (n_y z - n_z y)^2 + (n_z x - n_x z)^2 + R^2 > 0$$

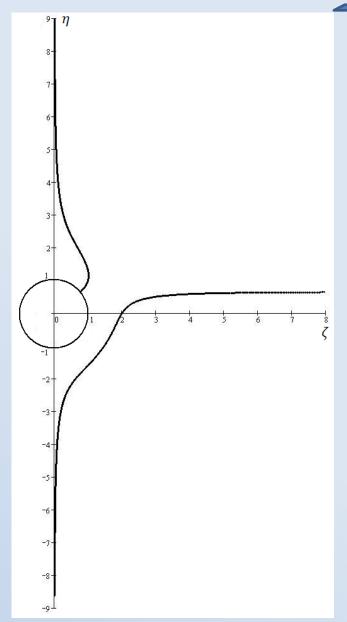
Thus both roots of this equation are real.

POSSIBLE EQUILIBRIA IN SPECIAL CASE



POSSIBLE EQUILIBRIA IN SPECIAL CASE





 $OP = (0, \sin(0.4), \cos(0.4))$

 $\mathbf{OP} = (0, \sin(0.7), \cos(0.7))$

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